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III/IV B.Tech (Regular) DEGREE EXAMINATION

July/August, 2023

Sixth Semester

Time: Three Hours

Mechanical Engineering

Heat Transfer

Maximum: 70 Marks

(14X1 = 14Marks)

(4X14=56 Marks)

Answer question 1 compulsory.

Answer one question from each unit.

Note: Heat and mass transfer data book is allowed.

- | | | | | |
|---|----|-----|----|---|
| | | CO | BL | M |
| 1 | a) | CO1 | L2 | 1 |
| | b) | CO1 | L2 | 1 |
| | c) | CO1 | L1 | 1 |
| | d) | CO2 | L2 | 1 |
| | e) | CO2 | L1 | 1 |
| | f) | CO2 | L1 | 1 |
| | g) | CO2 | L1 | 1 |
| | h) | CO3 | L2 | 1 |
| | i) | CO3 | L2 | 1 |
| | j) | CO3 | L2 | 1 |
| | k) | CO3 | L2 | 1 |
| | l) | CO4 | L1 | 1 |
| | m) | CO4 | L2 | 1 |
| | n) | CO4 | L1 | 1 |
- Why is the negative sign used in Fourier's law of heat conduction?
Differentiate between conductivity and conductance.
Define fin efficiency.
What is lumped system?
State the application of dimensional analysis in heat transfer processes.
Mention some of the areas where free and forced convection mechanisms are predominant.
What is the significance of Biot number?
What is the physical significance of Grash of number with reference to heat transfer by natural convection?
Differentiate between dropwise and filmwise condensation.
Distinguish between the pool boiling and flow boiling.
In a gas -to-liquid heat exchanger, why fins provided on the gas side?
Define emissivity.
What do you understand by monochromatic emissive power?
Give examples of some surfaces which do not appear black, but have high values of absorptivity
- Unit-I**
- | | | | | |
|---|----|-----|----|----|
| 2 | a) | CO1 | L2 | 7M |
| | b) | CO1 | L3 | 7M |
- Write down the expressions for the physical laws that govern each mode of heat transfer, and identify the variables involved in each relation.
A concrete wall 130 mm thick generates heat at the rate of 4000 w/m³ due to chemical reaction. Its surfaces are exposed to ambient air at a temperature of 20 °C. Calculate the surface temperature of wall and maximum temperature inside the wall. The thermal conductivity is 0.6 w/m K convection heat transfer co-efficient is 50 w/m² K.
- (OR)**
- | | | | | |
|---|----|-----|----|----|
| 3 | a) | CO1 | L3 | 7M |
| | b) | CO1 | L3 | 7M |
- Derive the general heat conduction equation in Cartesian coordinate system.
A composite wall is made up of two layer of 0.3m and 0.15m thickness respectively with outer surfaces of the composite wall held at 600°C and 20°C respectively. If conductivities are 20 and 50 W/mK, determine the heat conducted. In order to restrict the heat loss to 5 kW/m² another layer of 0.15m thickness is proposed. Determine the thermal conductivity of material required.
- Unit-II**
- | | | | | |
|---|----|-----|----|----|
| 4 | a) | CO2 | L3 | 7M |
| | b) | CO2 | L3 | 7M |
- A slab of aluminum 10cm thick is originally at a temperature of 500°C. It is suddenly immersed in a liquid at 100°C resulting a heat transfer coefficient of 1200 W/m²K. Determine the temperature at the centerline and the surface 1 minute after the immersion. Also calculate the total thermal energy removed per unit area of the slab during this period. The properties of aluminum for the given conditions are $\alpha = 8.4 \times 10^{-5} \text{ m}^2/\text{s}$, $K = 215 \text{ W/mK}$, $\rho = 2700 \text{ kg/m}^3$, $C = 0.9 \text{ kJ/kg-K}$.
Air at 30°C flows with a velocity of 10 m/s over a flat plate of length 1m and width 0.5 m whose surface is kept at a uniform temperature of 120°C. Determine the average heat transfer coefficient and the rate of heat transfer.

P.T.O

(OR)

- 5 a) A 12cm diameter long bar initially at a uniform temperature of 40°C is placed in a medium at 650°C with a convective coefficient of $22\text{W/m}^2\text{K}$. Calculate the time required for the bar to reach 255°C . Take $k = 20\text{W/m-K}$, $\rho = 580\text{kg/m}^3$ and $C = 1050\text{J/kg-K}$. CO2 L3 7M
- b) Engine oil at 25°C is forced over a $30\text{cm} \times 20\text{cm}$ plate at a velocity of 1.5m/s . The flow is parallel to the 30cm side of the plate, which is heated to a uniform temperature of 55°C . Calculate the rate of heat transfer from the plate to the oil. Properties of engine oil at 40°C are $\rho = 876\text{kg/m}^3$, $\nu = 24 \times 10^{-5}\text{m}^2/\text{s}$, $k = 0.144\text{W/mK}$ and $\text{Pr} = 2870$. CO2 L3 7M

Unit-III

- 6 a) Sketch temperature and velocity profiles in free convection on a vertical wall. CO3 L2 7M
- b) Derive an expression for LMTD in a parallel flow heat exchanger. CO3 L3 7M

(OR)

- 7 a) Draw the boiling curve for pool boiling of water and explain flow regime CO3 L2 7M
- b) In a lubricating oil cooler hot oil flowing at the rate of 2000kg/hr is cooled from 100°C to 60°C using cold water at 10°C flowing at the rate of 1500kg/hr . The specific heat of the oil is 2.6KJ/kg-K and water is 4.2KJ/kg-K . Determine the rate of heat transfer and outlet temperature of the cold fluid. CO3 L3 7M

Unit-IV

- 8 a) State Wien's displacement law and explain its significance CO4 L2 7M
- b) A black body is kept at a temperature of 1000K . Determine the emissive power of the body. CO4 L3 7M

(OR)

- 9 a) State and prove Kirchoff's law of radiation. CO4 L2 7M
- b) Emissivity of two large parallel plates maintained at 800°C and 300°C are 0.5 and 0.6 respectively. Find the percentage reduction in heat transfer when a polished aluminium radiation shield of emissivity 0.05 is placed between them. CO4 L3 7M



1a) $Q = -KA \left(\frac{dT}{dx} \right)$

As $\left(\frac{dT}{dx} \right)$ is negative, to give a +ve value for Q , negative sign is used.

b) $k \rightarrow$ Thermal Conductivity

$Q = \frac{\Delta T}{R} = K \cdot \Delta T \rightarrow K \rightarrow$ Thermal Conductance

Thermal Conductivity: — Ability to conduct heat
Defined as the ratio of HT rate to the temp. difference across the body.
Thermal Conductance: —
 \rightarrow Reciprocal of thermal resistance.

c) fin efficiency: — $\frac{Q_{fin}}{Q_{fin, max}} = \frac{\text{Actual HT from fin}}{\text{Ideal or max. HT from the fin if the entire fin is at the base temp.}}$

d) Lumped system: — At a given instant, the entire body is at the same temp. \Rightarrow There is no temp. gradient inside the body.

e) Dimensional analysis: — Convection HT.
Combining the variables into dimensionless groups, thereby analysis will be simplified.

f) Free Convection: — HT from I.C. engines, motors etc.

Forced Convection: — Cooling of electric components by fans
Cooling of oils and coolants in vehicles using forced air.

g) Biot no: $= \frac{hL_c}{k} = \frac{\text{Convection HT}}{\text{Conduction HT.}}$

$Bi < 0.1$, Represents lumped system.

h) Grashof no.: Criteria to predict laminar or turbulent flow.
 $Nu = f(Cr, Pr)$.

i) Dry Film Condensation: — Droplets form on surface and move downwards.

Filmwise Condensation: — A film forms on the surface.

Pool boiling: In this the vapour produced near the heating surface forms bubbles which grow and detach themselves from the surface and rise to the free surface due to buoyancy effect.

Flow boiling: In this case the boiling surface may itself be a portion of the flow passage. Generally associated with two phase flow confined through a passage.

Fins are provided on the gas side to minimize resistance.

Emissivity: - Ratio of emissive power of surface to emissive power of black body, at the same temp.

It is the emissive power at a given wavelength. (E_λ)

Snow, etc.

Unit-I

$Q = -KA \frac{dT}{dx}$ - Fourier's law - Conduction $\rightarrow 1M$

$Q = h \cdot A \cdot \Delta T$ - Newton's law of cooling - Convection $\rightarrow 1M$

$E_b = \sigma T^4$ - Stefan-Boltzmann's law - Radiation $\rightarrow 1M$

$Q = \text{Rate of HT}$

$K = \text{thermal conductivity}$

$A = \text{Area normal to surface}$

$\frac{dT}{dx} = \text{temp. gradient}$

$Q = \text{Rate of HT}$

$h = \text{Convection HT coefficient}$

$A = \text{Contact area or heat transfer area}$

$\Delta T = \text{temp. difference}$

$E_b = \text{Emissive power of black body}$

$\sigma = \text{Stefan Boltzmann's Const.}$

$T = \text{Surface temp. in } K$

Overall $\rightarrow 1M$

Conduction $\rightarrow 1M$

Convection $\rightarrow 1M$

Radiation $\rightarrow 1M$

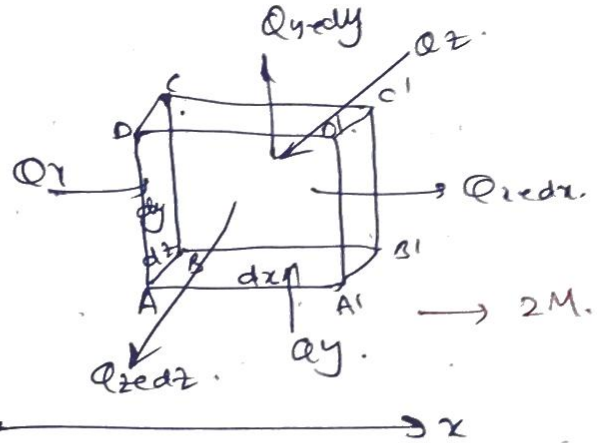
②

$$K_{wall} = 0.6 \text{ W/mK}; \quad h = 50 \text{ W/m}^2\text{K}$$

$$\tau_{\text{max}} = \tau_w e^{\frac{\eta L^2}{8\eta c}} = 25.2 + \frac{4000 \times (0.13)^2}{8 \times 0.6} = \underline{\underline{39.28^\circ\text{C}}} \rightarrow 34$$

General Hk equation.

$K_{x1}, K_2 \rightarrow$ Thermal conductivities
parallel along x & z direction



Energy balance of the solid element from 1st law of T.D.

Energy balance of the solid element from 1st law T.D.

$$\left[\text{Net heat conducted into the element per unit time} \right] + \left[\frac{\text{Internal heat generation Per unit time}}{(2)} \right] = \left[\begin{array}{l} \text{Change or increase in internal energy} \\ \text{Per unit time} \end{array} \right] + \left[\begin{array}{l} \text{Flow work done by the element} \\ \text{per unit time} \end{array} \right]$$

① ② ③ → 2M ④

Solid has no change is zero

Term IV : Negligible [Flow work done by solids due to temp change is zero]

Term iii : Change in internal energy during dt time

\therefore Change in internal energy / time = $P \cdot C_p \cdot \frac{dT}{dt} \cdot (d \times \text{area}) \rightarrow \text{IM}$

Term II: $= \dot{q}(dx dy dz)$ where \dot{q} = Internal H.C. per unit time per unit volume.

Term II: = $q(\alpha_2 dy dz)$ where α_2 is the heat flux in the y direction.

Term I: that will be conducted into and out of the element in each direction. We calculate net heat conducted for each direction and sum them to get the net heat conducted into the element per unit time.

x-direction $Q_x = -K_x \frac{\partial T}{\partial x} \cdot dxdz$

$$Q_{r+dr} = Q_r + \frac{\partial}{\partial r}(Q_r)dr = \frac{\partial}{\partial r}(k_r \frac{\partial T}{\partial r})(dr d\theta dz)$$

$$\therefore \text{Volumetric Conduction} = Q_n - Q_{n+dx} = - \frac{\partial}{\partial x} (k_y \frac{\partial T}{\partial y}) dx dy dz.$$

Y-direction \rightarrow Net heat conducted $= \frac{\partial}{\partial y} (k_y \frac{\partial T}{\partial y}) \cdot dx \cdot dz$
Z-direction \rightarrow Net heat conducted $= \frac{\partial}{\partial z} (k_z \frac{\partial T}{\partial z}) \cdot dx \cdot dy$

Substituting all terms and simplifying

$$\left(\frac{\partial}{\partial x} (k_x \frac{\partial T}{\partial x}) + \frac{\partial}{\partial y} (k_y \frac{\partial T}{\partial y}) + \frac{\partial}{\partial z} (k_z \frac{\partial T}{\partial z}) \right) dxdydz + \dot{q}(dxdydz) = \rho \cdot c_p \cdot \frac{\partial T}{\partial t} (dxdydz)$$

$$\div \text{ by } (dxdydz) \text{ and assuming isotropic material } \Rightarrow k_x = k_y = k_z = k.$$

$$\therefore \frac{\partial^2 T}{\partial x^2} + \frac{\partial^2 T}{\partial y^2} + \frac{\partial^2 T}{\partial z^2} + \frac{\dot{q}}{k} = \frac{\rho c_p}{k} \cdot \frac{\partial T}{\partial t} = \frac{1}{\alpha} \frac{\partial T}{\partial t} \rightarrow 1M$$

$$\textcircled{a} \quad \nabla^2 T + \frac{\dot{q}}{k} = \frac{1}{\alpha} \frac{\partial T}{\partial t}$$

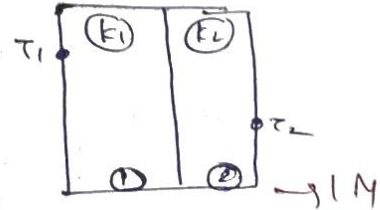
where α = Thermal diffusivity.

3b)

$$L_1 = 0.3 \text{ m}, \quad L_2 = 0.15 \text{ m}.$$

$$T_1 = 600^\circ\text{C}, \quad T_2 = 20^\circ\text{C}.$$

$$K_1 = 20 \text{ W/mK}, \quad K_2 = 50 \text{ W/mK}.$$

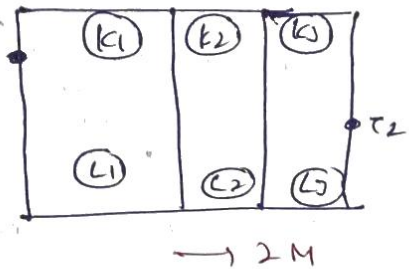


$$\frac{Q}{A} = \frac{T_1 - T_2}{\frac{L_1}{K_1} + \frac{L_2}{K_2}} = \frac{600 - 20}{\frac{0.3}{20} + \frac{0.15}{50}}$$

$$= \frac{580}{0.015 + 0.003} = \frac{580}{0.018} = 32222 \text{ W/m}^2 \rightarrow 3M.$$

$$Q = 5000 \text{ W/m}^2, \quad L_3 = 0.15 \text{ m}, \quad K_3 = ?$$

$$\frac{Q}{A} = \frac{T_1 - T_2}{\frac{L_1}{K_1} + \frac{L_2}{K_2} + \frac{L_3}{K_3}} = \frac{600 - 20}{\frac{0.3}{20} + \frac{0.15}{50} + \frac{0.15}{K_3}}$$



$$5000 = \frac{580}{0.015 + 0.003 + \left(\frac{0.15}{K_3}\right)} = \frac{580}{0.018 + \left(\frac{0.15}{K_3}\right)}$$

$$\therefore 0.018 + \left(\frac{0.15}{K_3}\right) = \frac{580}{5000} = 0.116$$

$$\therefore \frac{0.15}{K_3} = 0.116 - 0.018 = 0.098$$

$$\therefore K_3 = \frac{0.15}{0.098} = 1.53 \text{ W/mK}$$

$$\underline{K_3 = 1.53 \text{ W/mK}} \rightarrow 1M$$

4a)

$L_c = 10 \text{ cm} = 0.1 \text{ m}$ $T_0 = 500^\circ\text{C}$ $T_d = 100^\circ\text{C}$ $h = 1200 \text{ W/m}^2\text{K}$

$t = 60 \text{ ms}$ $T = 60$

$\alpha = 8.4 \times 10^{-5} \text{ m}^2/\text{s}$ $k = 215 \text{ W/mK}$ $\rho = 2700 \text{ kg/m}^3$

$C_p = 900 \text{ J/kgK}$

$L_c = \frac{L}{2} = \frac{0.1}{2} = 0.05 \text{ m} \rightarrow 2M$

$B_i = \frac{h L_c}{k} = \frac{1200 \times 0.05}{215} = 0.279 > 0.1 \rightarrow 2M$

\therefore Lumped system is not applicable $\rightarrow 3M$.

4b)

$V = 10 \text{ m/s} = u_d$ $L = 1 \text{ m}$ $W = 0.5 \text{ m}$ Flat plate -
Air.

$T_1 = 120^\circ\text{C}$ $T_2 = 30^\circ\text{C}$

$T_f = \frac{120 + 30}{2} = 75^\circ\text{C}$

properties of air at $T_f = 75^\circ\text{C}$ are

$\nu = 20.555 \times 10^{-6} \text{ m}^2/\text{s}$; $Pr = 0.693$ $k = 0.030065 \text{ W/mK} \rightarrow 1M$

$Re_L = \frac{u_d L}{\nu} = \frac{10 \times 1}{20.55 \times 10^{-6}} = 4.865 \times 10^5 < 5 \times 10^5 \rightarrow 2M$

\therefore Laminar flow - Flat plate - Constant wall temp. $\rightarrow 1M$

$\therefore \overline{Nu}_L = 0.664 \cdot Re_L^{0.5} \cdot Pr^{0.333}$
 $= 0.664 \times (4.865 \times 10^5)^{0.5} \cdot (0.693)^{0.333}$

$\frac{hL}{k} = 0.664 \times 697.5 \times 0.885 = 409.88$

$\therefore h = 409.88 \times \frac{k}{L} = \frac{409.88 \times 0.030065}{1} = 12.323 \text{ W/m}^2\text{K} \rightarrow 2M$

$Q = h \cdot A \cdot \Delta T = 12.323 \times (1 \times 0.5) \times (120 - 30)$
 $= 554.5 \text{ W} \rightarrow 1M$

5a)

$$T_0 = 40^\circ\text{C}, \quad T_\infty = 650^\circ\text{C}, \quad h = 22 \text{ W/m}^2\text{K}, \quad T_0 = 25^\circ\text{C} \quad (4)$$

$$\text{bar, } d = 12 \text{ cm} = 0.12 \text{ m} \Rightarrow R = 0.06 \text{ m}.$$

$$k = 20 \text{ W/mK}; \quad \rho = 580 \text{ kg/m}^3, \quad C_p = 1050 \text{ J/kgK}$$

$$L_c = \frac{R}{2} = \frac{0.06}{2} = 0.03 \text{ m} \quad \rightarrow 1 \text{ M}$$

$$Bi = \frac{h L_c}{k} = \frac{22 \times 0.03}{20} = 0.033 < 0.1 \quad \rightarrow 2 \text{ M}$$

\therefore Lumped system is applicable.

$$\therefore \frac{T - T_\infty}{T_0 - T_\infty} = \exp\left[-\frac{h A}{\rho C_p V} t\right] = \exp\left[-\frac{h}{\rho C_p L_c} t\right] \rightarrow 2 \text{ M}$$

$$\frac{25 - 650}{40 - 650} = \frac{395}{610} = 0.64754 = \exp\left[-\frac{h}{\rho C_p L_c} t\right]$$

Taking 'ln' on both sides.

$$\therefore -\frac{h}{\rho C_p L_c} t = \ln(0.64754) = -0.43457$$

$$\therefore \frac{h}{\rho C_p L_c} t = 0.43457$$

$$\therefore t = \frac{0.43457 \times \rho C_p L_c}{h} = \frac{0.43457 \times 580 \times 1050 \times 0.03}{22}$$

$$\therefore t = 361 \text{ seconds} \quad \rightarrow 2 \text{ M}$$

5b)

$$U_\infty = 1.5 \text{ m/s};$$

$$T_\infty = 25^\circ\text{C};$$

$$T_s = 55^\circ\text{C};$$

$$L = 0.2 \text{ m}, \quad W = 0.2 \text{ m};$$

$$\text{Properties: } \rho = 876 \text{ kg/m}^3$$

$$\text{at } T_f = 40^\circ\text{C}, \quad \mu = 24 \times 10^{-5} \text{ m}^2/\text{s}$$

$$k = 0.144 \text{ W/mK} \quad \rightarrow 1 \text{ M}$$

$$Pr = 2870$$

$$Re_L = \frac{U_\infty L}{\mu} = \frac{1.5 \times 0.2}{24 \times 10^{-5}} = 0.01875 \times 10^5 < 5 \times 10^5$$

\therefore Laminar flow. Flat plate. Constant wall temp. $\rightarrow 2 \text{ M}$

$$\therefore \overline{Nu}_L = 0.664 (Re_L)^{0.5} (Pr)^{0.333} = 0.664 \times (0.01875 \times 10^5)^{0.5} (2870)^{0.333} \quad \rightarrow 1 \text{ M}$$

$$\frac{h L}{k} = 0.664 \times 43.2 \times 14.1734 = 407.5$$

$$\therefore h = \frac{407.5 \times k}{L} = \frac{407.5 \times 0.144}{0.2} = 195.6 \text{ W/m}^2\text{K}$$

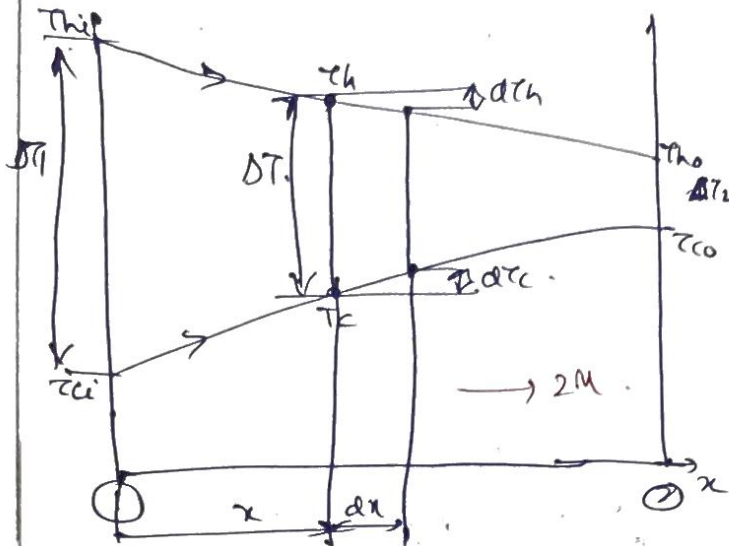
$$\therefore h = 195.6 \text{ W/m}^2\text{K} \quad \rightarrow 2 \text{ M}$$

$$\therefore Q = h \cdot A \cdot \Delta T = 195.6 \times (0.2 \times 0.2) \times (55 - 25)$$

$$Q = 352 \text{ Watts} \quad \rightarrow 1 \text{ M}$$

65D

CMTD for 11el flow HTE



Let us consider the HT between the hot and cold fluids for differential element of length dx

$dQ = U \cdot dA \cdot \Delta T = U \cdot dA (T_h - T_c) \rightarrow (1)$ where $dA = \text{width} \times dx$.
For the above differential element, the energy balance for the hot and cold fluids is

$dQ = -\dot{m}_h C_h dT_h = -\dot{C}_h dT_h \rightarrow (2)$ where $\dot{C}_h = \dot{m}_h C_h \rightarrow 1^{\text{st}} \text{ M.}$
 $dQ = \dot{m}_c C_c dT_c = \dot{C}_c dT_c$ where $\dot{C}_c = \dot{m}_c C_c$.

We know that $\Delta T = T_h - T_c \Rightarrow d(\Delta T) = dT_h - dT_c$

Substituting the values of dT_h and dT_c from (2).
 $-d(\Delta T) = \frac{-dQ}{\dot{C}_h} - \frac{dQ}{\dot{C}_c} = -dQ \left[\frac{1}{\dot{C}_h} + \frac{1}{\dot{C}_c} \right] \rightarrow 1^{\text{st}} \text{ M.}$
 $= -(U dA \cdot \Delta T) \left(\frac{1}{\dot{C}_h} + \frac{1}{\dot{C}_c} \right)$

Integrating over the entire length of heat exchanger,

$\int_{\Delta T_1}^{\Delta T_2} \frac{d(\Delta T)}{\Delta T} = (-U) \left(\frac{1}{\dot{C}_h} + \frac{1}{\dot{C}_c} \right) \int_0^L dA$

$= \ln \left(\frac{\Delta T_2}{\Delta T_1} \right) = (-UA) \left(\frac{1}{\dot{C}_h} + \frac{1}{\dot{C}_c} \right)$
 $= \frac{-UA}{Q} [(\dot{C}_h - \dot{C}_c) + (\dot{C}_c - \dot{C}_h)]$
 $= \left(\frac{-UA}{Q} \right) [(\dot{C}_h - \dot{C}_c) - (\dot{C}_h - \dot{C}_c)]$

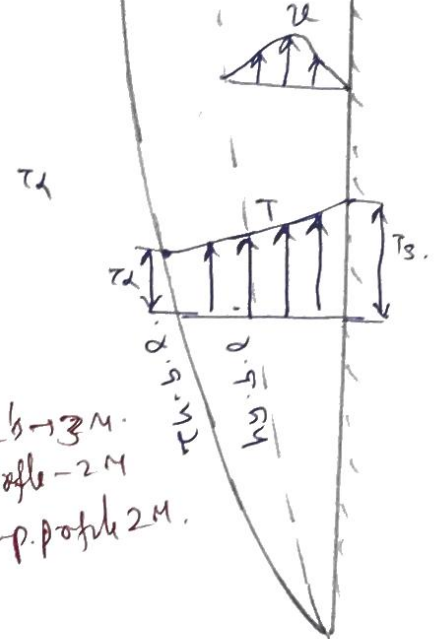
$\ln \left(\frac{\Delta T_2}{\Delta T_1} \right) = \frac{-UA}{Q} (\Delta T_1 - \Delta T_2) = \left(\frac{UA}{Q} \right) (\Delta T_2 - \Delta T_1)$

$\therefore Q = UA \cdot \left[\frac{\Delta T_2 - \Delta T_1}{\ln(\Delta T_2 / \Delta T_1)} \right] = \text{---}$

$Q = UA \cdot C LMTD$

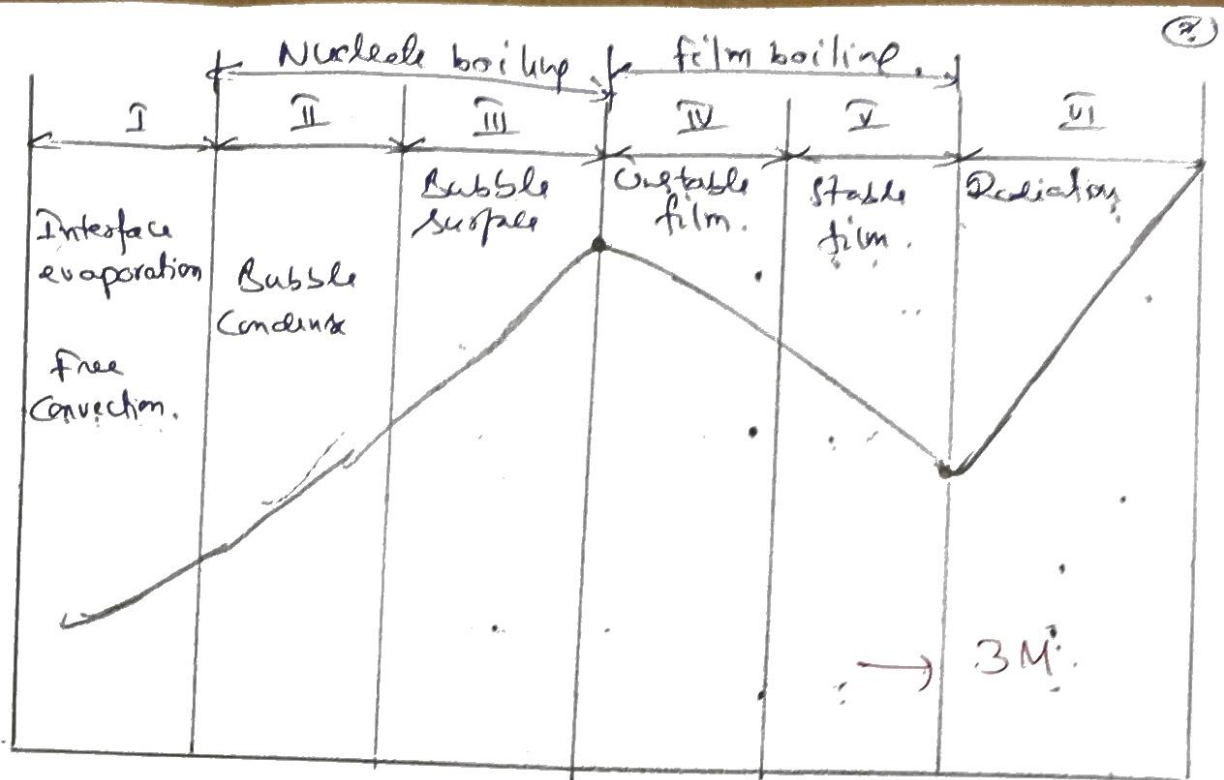
where $LMTD = \frac{\Delta T_2 - \Delta T_1}{\ln(\Delta T_2 / \Delta T_1)} \rightarrow 2^{\text{nd}} \text{ M.}$

66A



BC to 3M.
Up flow - 2M
Temp. profile 2M.

7a)



I → Interface evaporation

IV & III → Nucleate boiling

IV & V → Film boiling.

Brief explanation required.

→ 4M.

7b)

Hot fluid → oil

$$\dot{m}_h = 2000 \text{ kg/m.}$$

$$T_{hi} = 100^\circ\text{C.}$$

$$T_{ho} = 60^\circ\text{C}$$

$$C_h = 2600 \text{ J/kgK}$$

Cold fluid → water

$$\dot{m}_c = 1500 \text{ kg/m.}$$

$$T_{ci} = 10^\circ\text{C.}$$

$$C_c = 4200 \text{ J/kgK.}$$

$$Q = H.L. \text{ by hot fluid} = \dot{m}_h \cdot C_h \cdot (T_{hi} - T_{ho}) \rightarrow 1M$$

$$= \frac{2000}{3600} \times 2600 \times (100 - 60) = \underline{57777.8 \text{ Wats.}}$$

$$= \underline{57.777 \text{ kW}} \rightarrow 2M.$$

$$Q = H.C. \text{ by cold fluid}$$

$$= \dot{m}_c \cdot C_c \cdot (T_{co} - T_{ci})$$

$$\rightarrow 2M$$

$$\therefore (T_{co} - T_{ci}) = \frac{Q}{\dot{m}_c \cdot C_c} = \frac{57777.8}{\left(\frac{1500}{3600}\right) \times 4200} = 33^\circ\text{C}$$

$$\therefore T_{co} = 33 + T_{ci} = 33 + 10 = 43^\circ\text{C.}$$

$$\therefore \text{outlet temp. of water} = \underline{43^\circ\text{C.}} \rightarrow 2M$$

89) Wien's displacement law

$$\lambda_{\text{max}} \cdot T = \text{Constant} \rightarrow 3M$$

It is the relationship between temp. of a black body and the wavelength at which the maximum monochromatic emissive power occurs. $\rightarrow 3M$

$$E_{\lambda b} = \frac{C_1 \lambda^{-5}}{(e^{C_2/\lambda T} - 1)} \rightarrow 1M$$

$E_{\lambda b}$ becomes maximum if $\frac{d}{d\lambda}(E_{\lambda b}) = 0$
 \downarrow After differentiation and simplification.

$$\therefore \lambda_{\text{max}} \cdot T = \frac{C_2}{4.965} = \frac{1.4387 \times 10^{-2}}{4.965} \text{ mK}$$

$$\therefore \lambda_{\text{max}} \cdot T = 2898 \mu\text{m-K}$$

85) Black body, $T = 1000K$

$$E_b = \sigma T^4 = 5.67 \times 10^{-8} (1000^4)$$

$$\rightarrow 3M = 567 (10^4)$$

$$= 5.67 \times 10^4 \text{ W/m}^2$$

$$= 56700 \text{ W/m}^2 \rightarrow 4M$$

90) Kirchoff's law

It states that at any temp. the ratio of the total emissive power to the absorptivity is a constant for all substances which are in thermal equilibrium (with their environment).

② At thermal equilibrium, the absorptivity and emissivity of a body are equal. $\rightarrow 3M$

Proof: Let three bodies are in thermal equilibrium with each other, then

$$\frac{E_1}{\alpha_1} = \frac{E_2}{\alpha_2} = \frac{E_3}{\alpha_3} = \text{Const} \rightarrow 2M$$

If the third body is a black body, then

$$\frac{E_1}{\alpha_1} = \frac{E_2}{\alpha_2} = \frac{E_b}{\alpha_b = 1}$$

$$\therefore \frac{E_1}{\alpha_1} = E_b \Rightarrow \frac{E_1}{E_b} = \alpha_1 \Rightarrow E_1 = \alpha_1 E_b$$

$$\text{Similarly } \frac{E_2}{\alpha_2} = E_b \Rightarrow \frac{E_2}{E_b} = \alpha_2 \Rightarrow E_2 = \alpha_2 E_b$$

$$\therefore E = \alpha \cdot E_b \rightarrow 2M$$

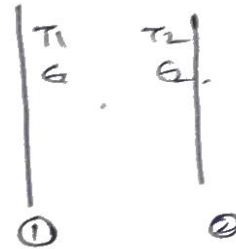
9.5)

Con 1 |.

$$T_1 = 800 + 273 = 1073 \text{ K}$$

$$T_2 = 300 + 273 = 573 \text{ K}$$

$$\epsilon_1 = 0.5, \epsilon_2 = 0.6$$



$$\begin{aligned} \therefore \left(\frac{Q_{12}}{A} \right) &= \frac{\sigma (T_1^4 - T_2^4)}{\frac{1}{\epsilon_1} + \frac{1}{\epsilon_2} - 1} \rightarrow 2 \text{ M} \\ &= \frac{5.67 \times 10^{-8} [1073^4 - 573^4]}{\left(\frac{1}{0.5} + \frac{1}{0.6} - 1 \right)} \\ &= \frac{5.67 (10.73^4 - 5.73^4)}{(2 + 1.667 - 1)} = \frac{5.67 (13255 - 1078)}{2.667} \\ &= 25889 \text{ W/m}^2 \rightarrow 2 \text{ M} \end{aligned}$$

$$\left(\frac{Q_{12}}{A} \right) = \frac{69046.4}{2.667} = 25889 \text{ W/m}^2 \rightarrow 2 \text{ M}$$

Con 2 |. A shield is placed between the plates.
 $\epsilon_s = 0.05$

$$\left(\frac{Q_{12}}{A} \right)_{\text{with shield}} = \frac{\sigma (T_1^4 - T_2^4)}{\left(\frac{1}{\epsilon_1} + \frac{1}{\epsilon_2} - 1 \right) + \left(\frac{2}{\epsilon_s} - 1 \right)} \rightarrow 3$$

$$= \frac{69046.4}{2.667 + 39} = \frac{69046.4}{41.667}$$

$$= 1657 \text{ W/m}^2 \rightarrow 2 \text{ M}$$

$$\begin{aligned} \% \text{ Reduction in HT} &= \left(\frac{25889 - 1657}{25889} \right) \times 100 \\ &= 93.6 \% \rightarrow 1 \text{ M} \end{aligned}$$